

# Choice of Design Variables

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1. Overview
2. Mesh Points
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# Choice of Design Variables

## Overview I

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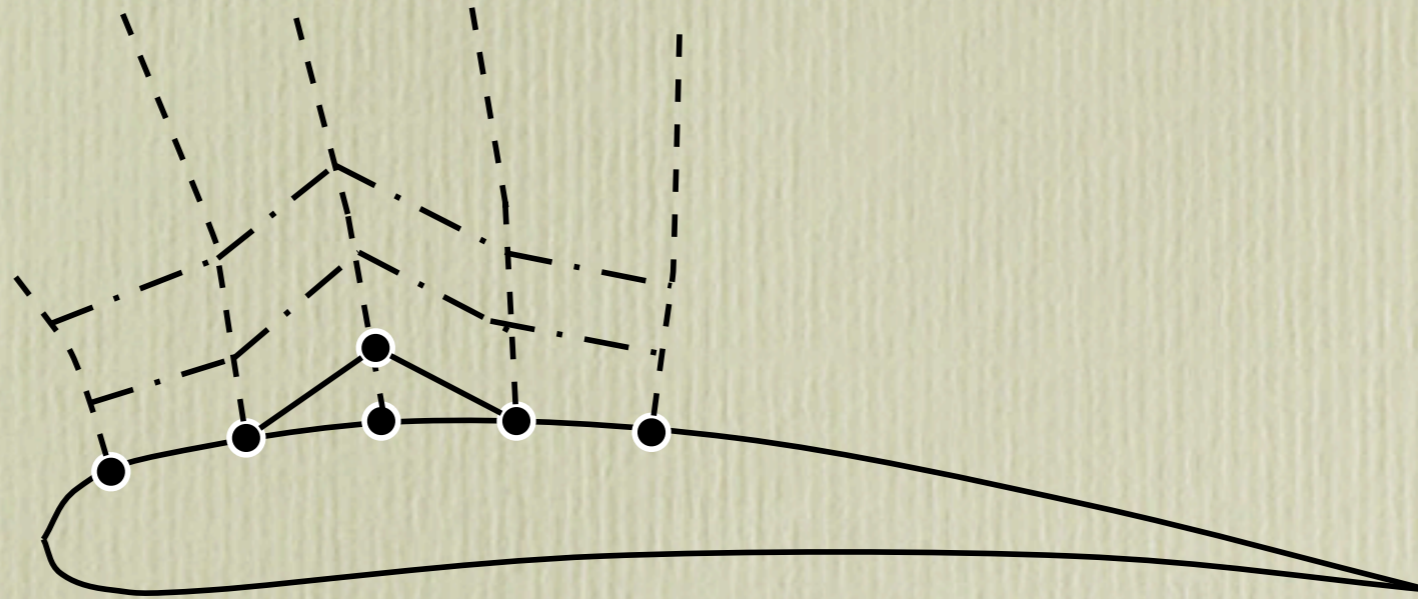
- Choice of design variables is one of the most important decisions in optimal design.
- Proper choice of design variables requires understanding of the flow physics and the type of design variables that affect the objective function.



## Choice of Design Variables

# Mesh Points I

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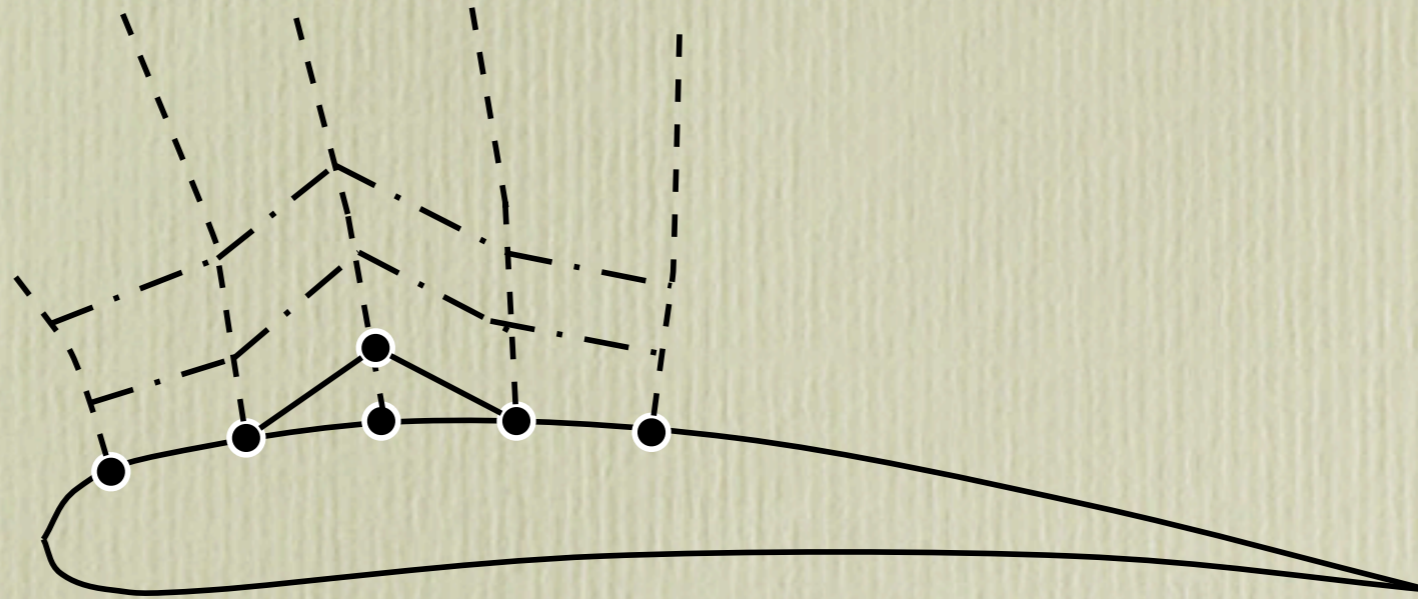
- Ensures that there is no restriction on the attainable geometry.
- If every surface mesh point is used, then complete design space containing the solution that achieves the global minimum is attainable.
- Gradients can be computed cheaply since  $\delta S_{i,j}$  is only non-zero for points along the perturbed grid line.



## Choice of Design Variables

# Mesh Points 2

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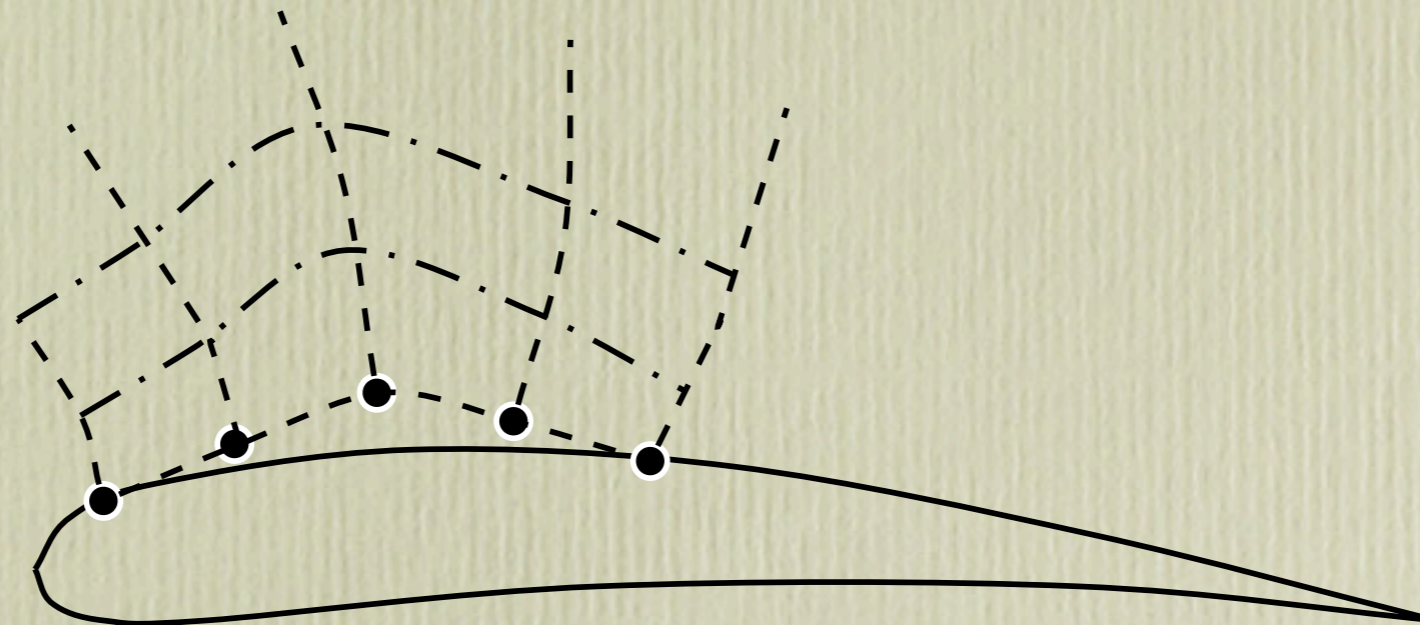
- If all surface mesh points are used, then method is **COSTLY** for 3D Viscous problems.
- Final geometry may not be smooth, due to gradients may contain high frequency modes.
- In an aerospace company, the final design stage requires the submission of the aircraft CAD lines. **BUT** CAD geometry may not match surface mesh geometry !!!



## Choice of Design Variables

# Hicks-Henne Functions I

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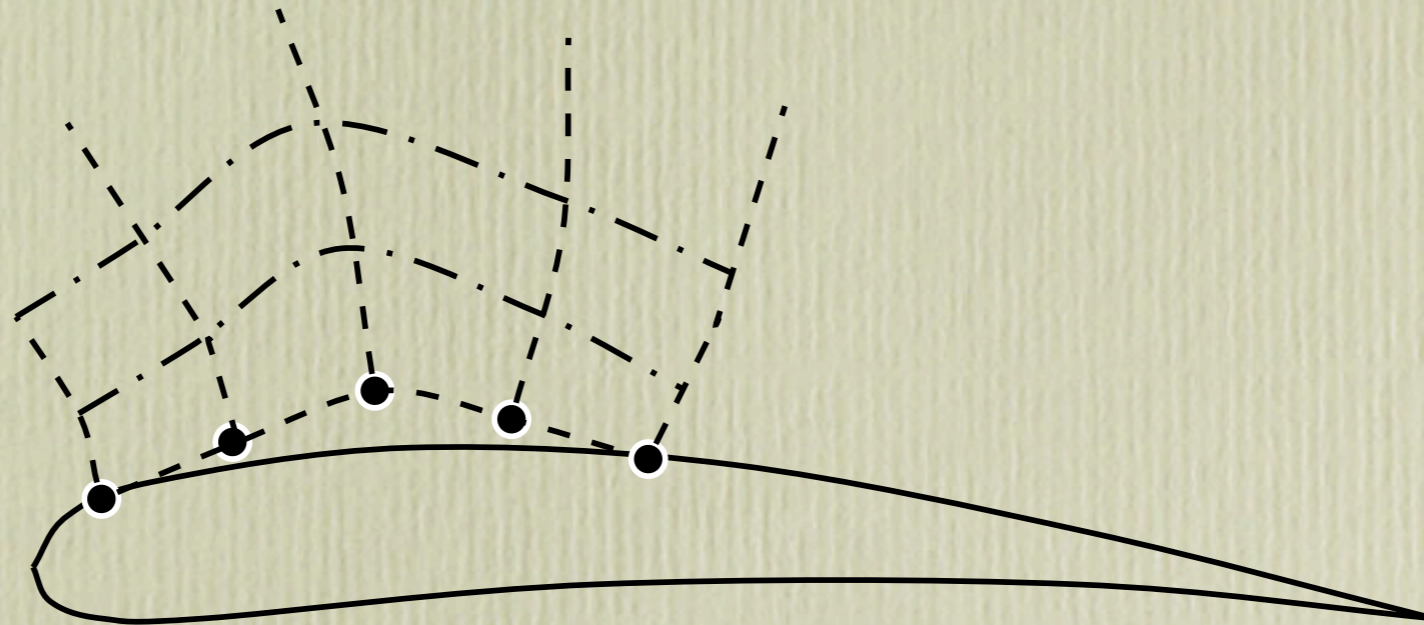
- Parameterize design space with set of smooth functions.
- Computed gradient remains smooth, and thus successive surface geometries in the design process remains smooth.
- Only a few Hicks-Henne sine bump functions are required to provide an acceptable span of the design space thus requiring fewer design variables.
- Bumps can be placed in the CAD geometry instead of the surface mesh, however, re-meshing can be costly.



# Choice of Design Variables

## Hicks-Henne Functions 2

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$$b(x) = a \left[ \sin \left( \pi x \frac{\log 5}{\log t_1} \right) \right]^{t_2} \quad \text{for } 0 \leq x \leq 1$$

- $a$  maximum bump magnitude
- $t_1$  locates the maximum point of the bump
- $t_2$  controls the width of the bump
- This flexibility allows one to place the bump at strategic points where a redesign is preferred while leaving other parts of the airfoil intact.